

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

A66EU
Revision 2
AVIONS PIERRE ROBIN
R 3000/160
May 4, 2001

TYPE CERTIFICATE DATA SHEET NO. A66EU

This data sheet which is part of Type Certificate No. A66EU prescribes conditions and limitations under which the Type Certificate was issued meets the airworthiness requirements of the Federal Aviation Regulations.

Type Certificate Holder: AVIONS PIERRE ROBIN
21121 Darois
France

I. Model R3000/160 Normal and Utility, approved November 17, 1988

Engine	Lycoming O-360 A3A		
Fuel	100 LL minimum aviation grade gasoline		
Engine Limits	For all operations 2600 rpm (160 hp)		
Propeller	Sensenich 76 EM8 S5-0-64		
Propeller Limits	Static rpm at maximum permissible throttle setting - 2200 ± 50 Diameter 76 inches. No cutoff permitted. Spinner ROBIN 58.26.10.1.80		
Airspeed Limits	Never exceed	314 km/h (170 kts)	
	Maximum structural cruising	249 km/h (134 kts)	
	Maneuvering	213 km/h (115 kts)	
	Flaps extended (35°)	174 km/h (94 kts)	
Flight Maneuvering Load Factors (G's)		<u>Cat. "U"</u>	<u>Cat. "N"</u>
	Flaps Up	+ 4.4	+ 3.8
		- 2.2	- 1.9
	Down	+ 2.0	+ 2.0
		0	0
C.G. Range	Forward limit 11.42 in. @ 1878 lbs. Intermediate limit 15.86 in @ 2205 lbs. in cat. "U" and 2536 lbs. in cat. "N". Aft limit 22.83 in. @ 2205 lbs. in cat. "U" and 2536 lbs. in cat. "N". Straight line variation between points given: 11.42 in. @ 1878 lbs. to 15.86 in. @ 2205 lbs. in cat. "U" and 2536 lbs. in cat. "N". 15.86 in. @ 2205 lbs. to 22.83 in. @ 2205 lbs. in cat. "U". 15.86 in. @ 2536 lbs. to 22.83 in. @ 2536 lbs. in cat. "N".		
Datum	Leading edge rectangular section.		
Leveling Means	Upper fuselage longeron		
Empty Weight C.G. Range	None		
Maximum Weight		<u>Cat. "U"</u>	<u>Cat. "N"</u>
	Takeoff	2205 lbs.	2536 lbs.
	Landing	2205 lbs.	2536 lbs.
No. of Seats	2 front at (19.68 - 23.62), 2 rear at (52.36)		

Page No.	1	2	3
Rev. No.	2	1	2

Maximum Baggage	88 lbs. at (+75). None for utility category.		
Fuel Capacity	66.19 US gal at (+9.84) (total) 59.66 US gal at (+9.84) (usable)		
Oil Capacity	Minimum Capacity	8 qts.	
	Minimum	2 qts.	
Control Surface Movements	Wing flaps	30° ± 2.0	
	Ailerons	Up 12° ± 2.0	Down 15° ± 2.0
	Rudder (L&R)	25° ± 3.0	
	Elevator	Up 25° ± 2.0	Down 10° ± 2.0
	Elevator trim tab (with elevator in neutral position)		
	Flaps retracted		
	Trim wheel to neutral	0° ± 3°	
	Trim wheel full down	- 12° ± 3°	
	Trim wheel full up	+ 37° ± 3°	
	Flaps extended (30°)		
	Trim wheel to neutral	+ 16° ± 3°	
	Trim wheel full down	- 9° ± 3°	
	Trim wheel full up	+ 40° ± 3°	
Serial Nos. Eligible	A French "Certificat de Navigabilité pour Exportation" endorsed as noted under "Import Requirements," must be submitted for each individual aircraft for which application for U.S. certification is made.		
Import Requirements	An FAA Standard Airworthiness Certificate may be issued on the basis of a French "Certificat de Navigabilité pour Exportation" signed by a representative of the Direction Générale de L'Aviation Civile (D.G.A.C.) of France, containing the following statement: "The airplane covered by this Certificate has been examined, tested, and found to conform to the type design approved under Type Certificate No. A66EU and to be in condition for safe operation."		
	For issuance of an airworthiness certificate in accordance with §21.183(c), the Direction Generale de l'Aviation Civile of France must certify that the aircraft conforms to the type design and is in condition for safe operation. In that regard, the Direction Generale de l'Aviation Civile of France will certify that the aircraft complies with the latest issue of Avions Pierre Robin Service Bulletins 111, 112, 114, 116, 121, 127, 132, 133, 136, 141, 146, 147, 151, 152, 153 and 176 and all other applicable mandatory continuing airworthiness information that it has issued, in addition to all other requirements of this type certificate data sheet. For issuance of an airworthiness certificate in accordance with §21.183(d), the certificating inspector, or other authorized person, must find, among other things, that the product is in a condition for safe operation.		
Certification Basis	Part 23 of the Federal Aviation Regulations, Amendments 1 through 23, FAR 21.29; FAR 36 Amendment 36-1 through (.....) Special Certification Review: "The Canopy (forward sliding) must be jettisonable." Date of applicable for type certification: October 26, 1990.		
Equipment	The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the aircraft for certification. In addition, the following is required: FAA-approved Airplane Flight Manual, Model R3000/160, Edition 2, April 1990.		

NOTE 1.

Current weight and balance report, including list of equipment in certificated empty weight, and loading instructions if applicable, must be in each aircraft at the time of original certification. Empty weight must include unusable fuel of 3.2 lbs. at (9.84). Basic empty weight includes engine oil of 16.5 lbs.

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